

Office of Naval Research
Department of the Navy
Contract Nonr-220(28)

STABILITY OF A CONDUCTING FLUID
FLOWING DOWN AN INCLINED PLANE
IN A MAGNETIC FIELD

by

Din-Yu Hsieh

Division of Engineering and Applied Science
CALIFORNIA INSTITUTE OF TECHNOLOGY
Pasadena, California

Report No. 85-29
December 1964

Approved by:
M. S. Plesset

Office of Naval Research
Department of the Navy
Contract Nonr-220(28)

STABILITY OF A CONDUCTING FLUID FLOWING
DOWN AN INCLINED PLANE IN A MAGNETIC FIELD

by

Din-Yu Hsieh

Reproduction in whole or in part is permitted
for any purpose of the United States Government

Division of Engineering and Applied Science
California Institute of Technology
Pasadena, California

Report No. 85-29
December, 1964

Approved by:
M.S. Plesset

STABILITY OF A CONDUCTING FLUID FLOWING DOWN AN INCLINED PLANE IN A MAGNETIC FIELD

Abstract

A stability analysis is made for the laminar flow of a layer of a viscous and electrically conducting fluid down an inclined plane in a transverse magnetic field. It is found that the effect of the magnetic field, revealed through the Hartmann number, is to stabilize the flow. A simpler and physically clearer approximate treatment of the same problem based on the principle of local balance is also given. The results agree quite satisfactorily with the exact analysis.

1. Introduction

The stability of laminar flows of an electrically conducting fluid in a magnetic field has been studied fairly extensively. Among others, Chandrasekhar⁽¹⁾ has investigated the stability of flow between coaxial rotating cylinders with a magnetic field in the axial direction, Stuart⁽²⁾ investigated the stability of pressure flow between parallel planes in a parallel magnetic field, while Lock⁽³⁾ has studied the latter stability with a magnetic field perpendicular to the direction of motion and to the boundary planes. In all these cases, it is found that the presence of magnetic fields tends to stabilize the system. In the present paper, we shall investigate the effect of a magnetic field on the stability of the gravity flow of a conducting fluid down an inclined plane. The latter stability for non-conducting fluids has been studied by Yih^{(4) (5)}, Benjamin⁽⁶⁾ and Binnie⁽⁷⁾, and also extended by the present author to the flow of superfluids⁽⁸⁾. It is found in general that the critical Reynolds number is quite low. From the study of the physical mechanism of this type of instability⁽⁹⁾, it may be inferred that

(1) S. Chandrasekhar, Proc. Roy. Soc. (London) A, 216, 293 (1953).

(2) J. T. Stuart, Proc. Roy. Soc. (London) A, 221, 189 (1954).

(3) R. C. Lock, Proc. Roy. Soc. (London) A, 233, 105 (1955).

(4) C.-S. Yih, Proc. 2nd U.S. Nat. Congr. Appl. Mech. (ASME, N.Y. 1955), 623.

(5) C.-S. Yih, Physics of Fluids, 6, 321 (1963).

(6) T. B. Benjamin, J. Fluid Mech. 2, 554 (1957).

(7) A. M. Binnie, J. Fluid Mech. 2, 551 (1957).

(8) D. Y. Hsieh, Physics of Fluids, 7, 1755 (1964).

(9) M. S. Plesset and D. Y. Hsieh, (in press).

the crucial feature is the velocity profile of the undisturbed flow. Therefore we may expect, as indeed can be verified by analysis, that a magnetic field parallel to the inclined plane will have relatively slight effects on the stability, while a transverse magnetic field may have quite pronounced effects. This expectation is also in agreement with the results of Stuart⁽²⁾ and Lock⁽³⁾. Therefore, in the following, we shall only consider the case in which the magnetic field is in the direction perpendicular to the inclined plane.

2. The Fundamental Equations

The hydromagnetic equations for a viscous, incompressible, conducting fluids are as follows.

Maxwell's equations:

$$\nabla \times \underline{H} = \frac{4\pi}{c} \underline{J} \quad , \quad (1)$$

$$\nabla \cdot \underline{H} = 0 \quad , \quad (2)$$

$$\nabla \times \underline{E} = - \frac{\mu}{c} \frac{\partial \underline{H}}{\partial t} \quad , \quad (3)$$

$$\epsilon \nabla \cdot \underline{E} = e \quad , \quad (4)$$

Ohm's law for a moving fluid:

$$\underline{J} = \sigma \left(\underline{E} + \frac{\mu}{c} \underline{v} \times \underline{H} \right) \quad , \quad (5)$$

The equation of continuity:

$$\nabla \cdot \underline{v} = 0 \quad (6)$$

The momentum equation:

$$\rho \frac{\partial \underline{v}}{\partial t} + \rho(\underline{v} \cdot \nabla) \underline{v} = -\nabla p + \frac{\mu}{c} (\underline{J} \times \underline{H}) + \eta \nabla^2 \underline{v} - \rho \nabla \Omega \quad (7)$$

and the energy equation

$$\frac{\partial U}{\partial t} + (\underline{v} \cdot \nabla) U = \frac{J^2}{\sigma} + \kappa \nabla^2 T + \frac{\eta}{2} \sum_{\alpha, \beta=1}^3 \left(\frac{\partial v_{\alpha}}{\partial x_{\beta}} + \frac{\partial v_{\beta}}{\partial x_{\alpha}} \right)^2, \quad (8)$$

where p , ρ , T and U are the pressure, density, temperature and internal energy of the fluid; ϵ , μ , σ , η and κ are the dielectric constant, permeability, coefficients of electrical conductivity, viscosity and thermal conductivity; Ω is the potential of the external force; and \underline{H} , \underline{E} , \underline{J} , and e are magnetic field, electric field, current density and charge density. In arriving at the above set of equations, we have assumed that the constitutive and transport coefficients are constant scalars, that the displacement current can be neglected in Eq. (1), that the force due to the electric field may be neglected in Eq. (7) and that the Ohm's law as stated by Eq. (5), which neglects the convection current, may be justified. We should also have the equation of state which gives the expression of U in terms of ρ and T to complete the system. However, as it stands, Eq. (8) is not coupled with other equations; hence it can be disregarded if no information about U and T is desired. Likewise, Eq. (4), which is also uncoupled from the other equations, only serves to determine the charge density e .

3. The Primary Flow

The primary flow to be considered is a layer of fluid flowing in parallel flow on a plane making an angle θ with the horizontal direction. An external uniform magnetic field H is applied in the direction perpendicular to the plane. Let the thickness of the layer be h . A coordinate

system is chosen with the origin at the free surface of the layer as shown in Fig. 1. Then Ω , the potential due to the gravitational force field, will be given by

$$\Omega = yg \cos \theta - xg \sin \theta \quad . \quad (9)$$

Now let us look for such an undisturbed system that

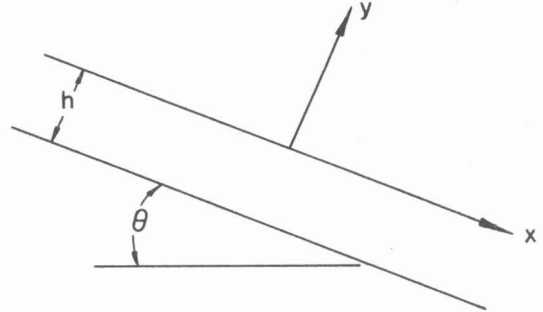


Figure 1

$$\underline{v} = (V(y), 0, 0) \quad , \quad \underline{H} = (Hb(y), H, 0) \quad , \quad \underline{J} = (0, 0, J(y)) \quad , \quad \underline{E} = (0, 0, E) \quad .$$

The electric field and current and additional magnetic field are introduced in order to satisfy the governing equations.

Equation (5) yields

$$J(y) = \sigma(E + \frac{\mu}{c} HV(y)) \quad . \quad (10)$$

Then Eq. (7) leads to $p = p(y)$, and

$$\eta V'' - \frac{\mu H \sigma}{c} \left(E + \frac{\mu H}{c} V \right) + \rho g \sin \theta = 0 \quad . \quad (11)$$

With the introduction of the Hartmann number $M = \frac{\mu H h}{c} \sqrt{\frac{\sigma}{\eta}}$, Eq. (11) can be written as

$$V'' - \frac{M^2}{h^2} V = \frac{\mu \sigma}{c \eta} HE - \frac{\rho g}{\eta} \sin \theta \quad . \quad (12)$$

As $V(-h) = V'(0) = 0$, we obtain

$$V(y) = V_0 \frac{\cosh M - \cosh \left(M \frac{y}{h} \right)}{\cosh M - 1} \quad , \quad (13)$$

where

$$V_o = \frac{(\cosh M - 1)h^2}{M^2 \cosh M} \left(\frac{\rho g}{\eta} \sin \theta - \frac{\mu \sigma}{c \eta} HE \right) . \quad (14)$$

The total discharge per unit width, incidentally, is

$$Q = \int_{-h}^0 V(y) dy = \frac{V_o}{\cosh M - 1} \left(h \cosh M - \frac{h}{M} \sinh M \right) , \quad (15)$$

while $\frac{dQ}{dh}$, if we use Eq. (14) and note the linear dependence on h of M , is found to be

$$\frac{dQ}{dh} = V_o \frac{\cosh M + 1}{\cosh M} . \quad (16)$$

We shall require that outside the fluid layer only the applied magnetic field $(0, H, 0)$ is present, and that the total flux of current in the z -direction is zero. The latter requirement leads to the condition

$$\int_{-h}^0 J(y) dy = 0 \quad (17)$$

while the former requires that

$$b(0) = b(-h) = 0 . \quad (18)$$

Equations (10), (13) and (17) then lead to

$$E = \frac{\mu H V_o}{c} \frac{\sinh M - M \cosh M}{M(\cosh M - 1)} , \quad (19)$$

and then Eq. (5) yields

$$J(y) = \frac{\sigma \mu H V_o}{c} \frac{\sinh M - M \cosh \left(M \frac{y}{h} \right)}{M(\cosh M - 1)} . \quad (20)$$

Equation (1), with Eq. (18), then leads to

$$b(y) = - \frac{4\pi\sigma\mu V_0}{c^2} \frac{y \sinh M - h \sinh\left(\frac{My}{h}\right)}{M(\cosh M - 1)}.$$

4. The Stability Problem

Let us superimpose a two-dimensional disturbance on the primary flow so that the elevation of the free surface is given by

$$y = \zeta(x, t) = \zeta e^{ik(x-at)} \quad (22)$$

Thus any perturbed quantity $g(x, y, t)$ may be written as

$$g(x, y, t) = g(y) e^{ik(x-at)} \quad (23)$$

After the system is perturbed, we have $\underline{v} = (v_x, V+v_y, v_z)$,

$\underline{H} = (Hb+H_x, H+H_y, H_z)$, $\underline{J} = (J_x, J_y, J+J_z)$ and $\underline{E} = (E_x, E_y, E+E_z)$. We

shall assume that the disturbance is small and that the squares of the small quantities in all the equations can be neglected. Eliminating p

from Eq. (7) and using Eqs. (1), (2) and (6), we obtain

$$\begin{aligned} v_y^{iv} - 2k^2 v_y''' + k^4 v_y'' - ik \frac{\rho}{\eta} [(V-a)(v_y'' - k^2 v_y') - V'' v_y'] \\ = \frac{ik\mu H}{4\pi\eta} \left[\frac{i}{k} (H_y''' - k^2 H_y'') - b(H_y'' - k^2 H_y') + b'' H_y' \right] \end{aligned} \quad (24)$$

We can make the above equation non-dimensional by taking $v_y = V_0 \psi$,

$H_y = H_0 \phi$, $V = V_0 \bar{V}$, $a = V_0 \bar{a}$, $k = \frac{\bar{k}}{h}$ and $y = h\bar{y}$, and substitute them into the equation. Dropping bars of the newly introduced quantities to simplify writing, we find that Eq. (24) becomes

$$\begin{aligned}
& [\psi^{IV} - 2k^2\psi'' + k^4\psi] - ikR [(V-a)(\psi'' - k^2\psi) - V''\psi] \\
& = - \frac{M^2}{R_M} [\varphi''' - k^2\varphi' + ikb(\varphi' - k^2\varphi) + ikb''\varphi] \quad , \quad (25)
\end{aligned}$$

where

$$R = \frac{V_o \rho h}{\eta} \quad , \quad (26)$$

and

$$R_M = \frac{4\pi\mu\sigma V_o h}{c^2} \quad . \quad (27)$$

R and R_M are called the Reynolds number and magnetic Reynolds number respectively. Eliminating J_z from Eqs. (1) and (5), and using Eqs. (2) and (3), we obtain similarly:

$$\varphi'' - k^2\varphi = -R_M [\psi' + ikb\psi - ik(V-a)\varphi] \quad . \quad (28)$$

The electromagnetic fields in regions outside the fluid layer are governed by the Maxwell equations. With the space and time variation given by Eq. (23), the field quantities can be readily determined for a medium which is conducting or non-conducting.

The boundary conditions will contain the following features.

(i) The normal velocity of the fluid vanishes at the bottom:

$$\psi(-1) = 0 \quad . \quad (29)$$

(ii) The tangential velocity of the fluid vanishes at the bottom:

$$\psi'(-1) = 0 \quad , \quad (30)$$

after use of Eq. (6).

(iii) The shearing stress vanishes at the free surface;

$$\frac{\partial}{\partial x} (v_y) + \frac{\partial}{\partial y} (V + v_x) = 0 \quad \text{on} \quad y = \zeta(x, t) ;$$

or

$$\psi''(0) + k^2 \psi(0) + \frac{ikM^2}{\cosh M - 1} \zeta = 0 . \quad (31)$$

(iv) The normal stress is continuous across the free surface:

$$p = 2\eta \left(\frac{\partial v_y}{\partial y} \right) - \Gamma \rho \left(\frac{\partial^2 \zeta}{\partial x^2} \right) + \rho g \cos \theta \zeta , \quad \text{at} \quad y = 0 ,$$

where $\rho\Gamma$ is the surface tension coefficient. Using Eq. (7), we then obtain

$$\begin{aligned} (\psi''' - k^2 \psi') + \frac{M^2}{R_M} (\varphi'' - k^2 \varphi) - ik \{ R[(V-a)\psi' - V'\psi] + \frac{M^2}{R_M} b'\varphi \} \\ = 2k^2 \psi' + k^2 \left(\Gamma k^2 R + \frac{M \sinh M}{\cosh M - 1} \cot \theta \right) \zeta , \quad \text{at} \quad y = 0, \quad (32) \end{aligned}$$

where the relation $V_0 = \frac{\rho g \sin \theta h^2}{\eta} \frac{\cosh M - 1}{M \sinh M}$ which is derivable from Eqs. (14) and (19) has been used.

(v) The free surface will satisfy the kinematic surface condition:

$$\psi(0) - ik(V_0 - a)\zeta = 0 . \quad (33)$$

(vi) The electromagnetic fields in the fluid layer will be appropriately connected to those in the free space above and those in the material wall below.

5. Solution for Long Waves

In the analysis of the stability problem for non-conducting fluids⁽⁵⁾, it is shown that the criterion for stability is essentially determined by the

disturbances of long wavelength. We expect this feature to remain true so long as the magnetic field is not too strong. When the wavelength of the disturbance is long compared with the depth of the fluid layer, i.e., when $k \ll 1$, a scheme of successive approximation can be developed. For the first approximation, we shall neglect all quantities of order $O(k)$ in the system of differential equations and boundary conditions. Then Eqs. (25) and (28) become

$$\psi_o^{iv} + \frac{M^2}{R_M} \varphi_o''' = 0 \quad , \quad (34)$$

and

$$\varphi_o'' + R_M \psi_o' = 0 \quad , \quad (35)$$

and the boundary conditions become

$$(i) \quad \psi_o(-1) = 0 \quad , \quad (36)$$

$$(ii) \quad \psi_o'(-1) = 0 \quad , \quad (37)$$

$$(iii) \quad \psi_o''(0) + \frac{M^2}{\cosh M - 1} ik \zeta_o = 0 \quad , \quad (38)$$

$$(iv) \quad \psi_o'''(0) + \frac{M^2}{R_M} \varphi_o''(0) = 0 \quad , \quad (39)$$

and

$$(v) \quad \psi_o(0) - (V_o - a_o) ik \zeta_o = 0 \quad , \quad (40)$$

where the subscript zero signifies that it is the solution for the first approximation.

Equations (34) and (35) lead to

$$\psi_o^{iv} - M^2 \psi_o'' = 0 \quad . \quad (41)$$

Thus the general solution is

$$\psi_o = A_o \cosh My + B_o \sinh M_y + C_o y + D_o \quad . \quad (42)$$

Now Eq. (39), with application of Eq. (35), becomes

$$\psi_o'''(o) - M^2 \psi_o'(o) = 0$$

which requires

$$C_o = 0 \quad .$$

Then Eqs. (36) and (37) lead to

$$\psi_o = A_o [\cosh M(1+y) - 1] \quad . \quad (43)$$

From Eq. (38), we get

$$ik\zeta_o = -A_o \cosh M(\cosh M - 1) \quad , \quad (44)$$

and from Eq. (40), as $V_o = 1$, we obtain

$$a_o = \frac{\cosh M + 1}{\cosh M} \quad . \quad (45)$$

It may be noted that a_o is identical with the expression of $\frac{dQ}{dh}$ given by Eq. (16).

The general expression for φ_o can be written down immediately from Eq. (35):

$$\varphi_o = -R_M \left[\frac{A_o}{M} \sinh M(1+y) + C_o' y + D_o' \right] \quad . \quad (46)$$

The coefficients C_o' and D_o' can be readily determined, by the application of the boundary conditions (vi). Now for almost all conducting liquid of interest, the ratio $\frac{R_M}{R}$ is extremely small. For mercury, it is about 1.5×10^{-7} , while for liquid sodium, it is 7.5×10^{-6} . Therefore in the range of Reynolds number usually encountered in laminar flow, R_M is

very small. The value of φ_0 , as may be seen easily, is in general of the order $O(R_M)$. Hence, as far as the stability of the flow is concerned, there is practically no need to obtain the explicit expression for φ_0 .

For the next approximation, we shall take $\psi = \psi_0 + \psi_1$, where $\psi_1 = O(k)$, etc., and neglect all terms that are quadratic in k in the differential equations and boundary conditions. Keeping in mind that $\varphi_0 = O(R_M)$ and $b = O(R_M)$, we obtain from Eqs. (25) and (28):

$$\psi_1^{IV} - M^2 \psi_1'' = ik \{ R [(V - a_0) \psi_0'' - V'' \psi_0] + O(R_M) \} \quad (47)$$

As

$$V = \frac{\cosh M - \cosh My}{\cosh M - 1}, \quad V'' = - \frac{M^2 \cosh My}{\cosh M - 1},$$

thus with $A_0 = 1$, Eq. (47) becomes

$$\psi_1^{IV} - M^2 \psi_1'' = ikR \frac{M^2 \tanh M \sinh My}{\cosh M - 1} \quad (48)$$

The general solution of the above equation is:

$$\psi_1 = A_1 \cosh My + B_1 \sinh My + C_1 y + D_1 + \frac{ikR}{2M} \frac{\tanh M}{\cosh M - 1} y \cosh My \quad (49)$$

Now, from Eq. (28), we obtain:

$$\varphi_1'' + R_M \psi_1' = O(R_M^2) \quad (50)$$

Thus in particular,

$$\varphi_1''(0) = -R_M \left(B_1 M + C_1 + \frac{ikR}{2M} \frac{\tanh M}{\cosh M - 1} \right) \quad (51)$$

The boundary condition (iv) becomes

$$\psi_1'''(0) + \frac{M^2}{R_M} \psi_1''(0) = ik \left\{ R[(V_0 - a_0)\psi_0'(0) - V'(0)\psi_0(0)] - ik\zeta_0 \left[\Gamma k^2 R + \frac{M \sinh M}{\cosh M - 1} \cot \theta \right] + O(R_M) \right\} \quad (52)$$

Then Eqs. (49) and (51) together with the first order solutions lead to

$$C_1 = ik \left\{ R \frac{\sinh M}{M(\cosh M - 1)} - \frac{\cosh M}{M^2} [M \sinh M \cot \theta + \Gamma k^2 R(\cosh M - 1)] \right\}. \quad (53)$$

Boundary conditions (i) and (ii) yield

$$A_1 \cosh M - B_1 \sinh M - C_1 + D_1 = \frac{ikR}{2M} \frac{\sinh M}{\cosh M - 1}, \quad (54)$$

and

$$-A_1 M \sinh M + B_1 M \cosh M + C_1 = -\frac{ikR}{2M} \frac{\tanh M(\cosh M + M \sinh M)}{\cosh M - 1}, \quad (55)$$

while boundary conditions (iii) and (v) are

$$\psi_1''(0) + \frac{M^2}{\cosh M - 1} ik\zeta_1 = 0, \quad (56)$$

and

$$\psi_1(0) - ik\zeta_1(V_0 - a_0) + ik\zeta_0 a_1 = 0, \quad (57)$$

and they combine to give

$$a_1 = \frac{A_1 + D_1 \cosh M}{\cosh^2 M(\cosh M - 1)}. \quad (58)$$

Multiply Eq. (54) by $\cosh M$, Eq. (55) by $\frac{\sinh M}{M}$, and then add to obtain

$$A_1 + D_1 \cosh M = C_1 \left(\cosh M - \frac{\sinh M}{M} \right) + \frac{ikR \sinh M}{2M(\cosh M - 1)} \left[\cosh M - \frac{\sinh M}{M} - \frac{\sinh^2 M}{\cosh M} \right] \quad (59)$$

Using Eq. (53), we then arrive at:

$$a_1 = \frac{ik}{\cosh^2 M (\cosh M - 1)} \left\{ R \frac{\tanh M}{2M(\cosh M - 1)} \left(3\cosh^2 M - \frac{3\cosh M \sinh M}{M} - \sinh^2 M \right) - \frac{\cosh M}{M^2} \left(\cosh M - \frac{\sinh M}{M} \right) [M \sinh M \cot \theta + (\cosh M - 1) \Gamma k^2 R] \right\} . \quad (60)$$

6. Stability Criteria

The flow system is stable or unstable according as the imaginary part of $a = a_0 + a_1$ is positive or negative. For this problem, a_0 is real, while a_1 is imaginary; therefore, the stability criterion is determined by the sign of a_1 .

It may be seen from the expression of a_1 , that coefficients associated with Γ and $\cot \theta$ are all negative; this means that surface tension always tends to stabilize the flow while gravity will tend to stabilize or destabilize the system according to whether the fluid is flowing down the upperside or the underside of the plane.

If we neglect the effect of surface tension, we may conclude that the flow is stable if

$$R < F(M) \cot \theta , \quad (61)$$

where

$$F(M) = \frac{2 \cosh^2 M \left(\cosh M - \frac{\sinh M}{M} \right) (\cosh M - 1)}{3 \cosh^2 M - \frac{3 \cosh M \sinh M}{M} - \sinh^2 M} . \quad (62)$$

$F(M)$ is a monotonically increasing function and as a function of M is shown in Fig. 2. Thus we may conclude that the magnetic field tends to

stabilize the flow system. As $M \rightarrow 0$, it may be readily verified that $F(M) \rightarrow \frac{5}{4}$, thus the results agree with those of Yih⁽⁵⁾ and Benjamin⁽⁶⁾. For large M ,

$$F(M) \approx \frac{M-1}{2(2M-3)} e^{2M}. \quad (63)$$

The above result will not hold for very large M , since in that case, the stability will be most likely controlled by the shear wave disturbance rather than these soft waves⁽⁵⁾, while Lock⁽³⁾ has shown that the critical Reynolds number for the former type of disturbance in his problem is only linearly proportional to M for large M .

7. Simple Description of the Stability Problem

The principle of local balance⁽¹⁰⁾ has been extended to the stability problem of laminar flow down an inclined plane⁽⁹⁾. That approach offers a much simpler analysis and clearer physical picture while it retains satisfying accuracy. We shall also apply that principle to this problem.

Once the primary flow is obtained, the speed of propagation of disturbances of long wavelength, a_o , may be immediately written down from the total discharge as expressed in Eq. (16)⁽¹¹⁾. Thus

$$a_o = \frac{dQ}{dh} = V_o \frac{\cosh M+1}{\cosh M}. \quad (64)$$

Let us now transform the coordinate system from the original (x, y) to a new coordinate system (x', y) by the Galilean transformation:

⁽¹⁰⁾ M.S. Plesset and D.Y. Hsieh, Physics of Fluids, 7, 1099 (1964).

⁽¹¹⁾ M.J. Lighthill and G.B. Whitham, Proc. Roy. Soc. A, 229, 281 (1955).

$$x' = x - a_0 t \quad , \quad y' = y \quad , \quad (65)$$

which brings the wave disturbance of the free surface to rest. In this new frame of reference, the disturbance wave is given by

$$y = \zeta e^{ikx'} \quad , \quad (66)$$

or, taking the imaginary part:

$$y = \zeta \sin kx' \quad . \quad (66')$$

Also in the new frame, the fluid in the layer is moving with velocity:

$$U(y) = V(y) - a_0 = V_0 \frac{1 - \cosh M \cosh(My/h)}{\cosh M(\cosh M - 1)} \quad . \quad (67)$$

We now compute the pressure exerted on the "wavy wall" (Eq. (66')) by a layer of incompressible inviscid fluid bounded by a rigid wall at $y = -h$, with primary velocity given by Eq. (67). Let the fluid velocity in the presence of a wavy wall be given by $(U+u, v)$, then the momentum and continuity equation for steady flow are

$$(U + u) \frac{\partial u}{\partial x} + v \frac{\partial}{\partial y} (U + u) = - \frac{1}{\rho} \frac{\partial p}{\partial x} \quad , \quad (68)$$

$$(U + u) \frac{\partial v}{\partial x} + v \frac{\partial v}{\partial y} = - \frac{1}{\rho} \frac{\partial p}{\partial y} \quad , \quad (69)$$

and

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0 \quad . \quad (70)$$

We eliminate p and u and neglect quadratic terms in u and v , and get

$$\frac{\partial^2 v}{\partial y^2} + \frac{\partial^2 v}{\partial x^2} - \frac{U''}{U} v = 0 \quad . \quad (71)$$

Now let

$$v(x, y) = f(y) \cos kx \quad . \quad (72)$$

Then Eq. (71) becomes

$$f'' - \left(\frac{U''}{U} + k^2 \right) f = 0 \quad . \quad (73)$$

For long wavelength disturbance, we have $kh \ll 1$, thus we may approximate Eq. (73) by

$$f'' - \frac{U''}{U} f = 0 \quad . \quad (74)$$

One solution of the last equation is U , then by reduction of order, we obtain the general solution of Eq. (74):

$$f = A U(y) + B U(y) \int_0^y \frac{dy'}{U^2(y')} \quad . \quad (75)$$

The boundary conditions at the wavy surface and the bottom require that:

$$f = 0 \quad \text{at} \quad y = -h \quad , \quad (76)$$

and

$$f = U_0 k \zeta \quad \text{at} \quad y = 0 \quad , \quad (77)$$

where

$$U_0 = U(0) \quad .$$

Thus

$$f = k \zeta \left[U(y) + \frac{U(y)}{G} \int_0^y \frac{dy'}{U^2(y')} \right] \quad . \quad (78)$$

where

$$G = \int_0^h \frac{dy}{U^2(y)} \quad . \quad (79)$$

For this case, as $U'(0) = 0$, the pressure applied to the wavy wall is readily seen from Eqs. (68) and (70) to be:

$$p(0) = \frac{\rho U_0 f'(0)}{k} \sin kx \quad . \quad (80)$$

As

$$f'(y) = k\zeta \left[U'(y) + \frac{U'(y)}{G} \int_0^y \frac{dy'}{U^2(y')} + \frac{1}{GU(y)} \right] ,$$

we obtain

$$p(0) = \frac{\rho\zeta}{G} \sin kx \quad . \quad (81)$$

This pressure is counterbalanced by the gravitational restoring force $\rho g \cos \theta \zeta \sin kx$ in the stability problem, (cf. Fig.1), hence the critical condition for stability may be expressed by

$$g \cos \theta = \frac{1}{G} \quad . \quad (82)$$

Now

$$\begin{aligned} G &= \int_0^h \frac{dy}{U^2(y)} = \frac{\cosh^2 M (\cosh M - 1)^2}{V_0^2} \int_0^h \frac{dy}{\left(\cosh M \cosh M \frac{y}{h} - 1 \right)^2} \\ &= \frac{\cosh^2 M (\cosh M - 1)^2 h}{V_0^2 M} \int_0^M \frac{dz}{(\cosh M \cosh z - 1)^2} . \end{aligned}$$

The integral is readily integrated and we obtain

$$\frac{1}{G} = \frac{V_0^2 M}{\cosh^2 M (\cosh M - 1)^2 h} \frac{\sinh^3 M}{\cosh M + \pi/2} \quad . \quad (83)$$

As in the derivation of Eq. (32) we can now express

$$g \cos \theta = \frac{V_0 \eta \cot \theta}{\rho h^2} \frac{M \sinh M}{\cosh M - 1} \quad . \quad (84)$$

Then Eq. (82) may be expressed as

$$\frac{V_o \eta \cot \theta}{\rho h^2} \frac{M \sinh M}{\cosh M - 1} = \frac{V_o^2 M \sinh^3 M}{h \cosh^2 M (\cosh M - 1)^2 (\cosh M + \pi/2)} ,$$

or

$$F'(M) \cot \theta = R , \quad (85)$$

where

$$F'(M) = \frac{\cosh^2 M (\cosh M - 1) (\cosh M + \pi/2)}{\sinh^2 M} . \quad (86)$$

The comparison between $F(M)$ and $F'(M)$ is shown in Fig. 2. The general agreement between F and F' over all values of M shows that the above simple description reveals the essential physical mechanism of the instability.

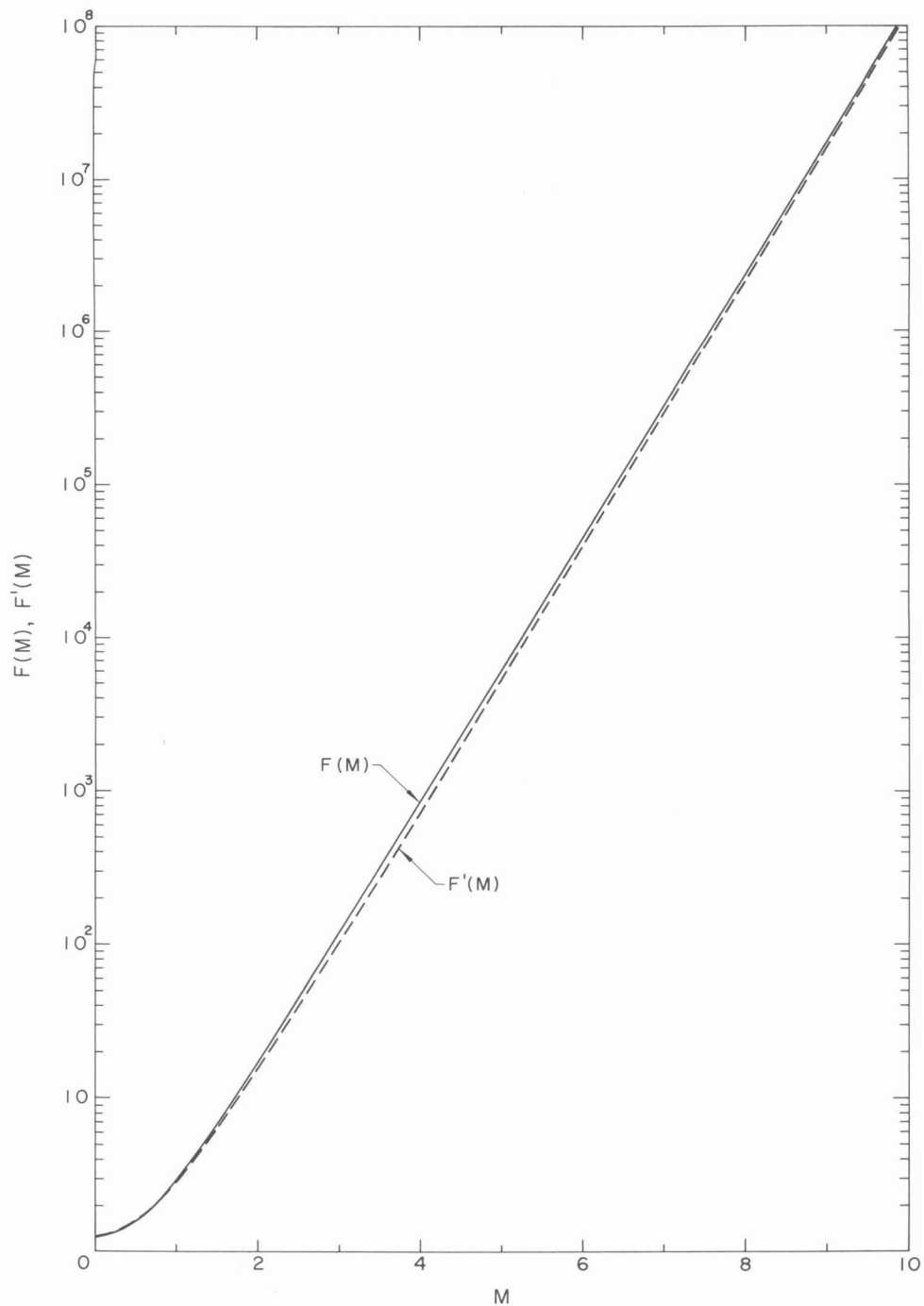


Figure 2. Comparison of the critical values of $R/\cot \theta$, as a function of the Hartmann number M , where R is the Reynolds number, and θ is the angle of inclination of the plane; F is the value for this ratio obtained from the complete theory, and F' is that obtained from the approximate theory of local balance.

DISTRIBUTION LIST FOR UNCLASSIFIED TECHNICAL REPORTS ISSUED UNDER

Contract Nonr-220(28) Task Nr 062-059
Single Copies Unless Otherwise Given

Chief of Naval Research Dept. of the Navy Washington 25, D. C. Attn: Codes 438 (3) 461 463 429	Chief, Bureau of Ships Department of the Navy Washington 25, D.C. Attn: Codes 300 305 335 341 342A 345 421 440 442 634A	Hydrographer U.S. Navy Hydrographic Off. Washington, 25, D.C. Commanding Off. and Dir. U. S. Navy Engineering Laboratory Annapolis, Maryland Attn: Code 750 Commander U. S. Naval Weapons Lab. Dahlgren, Virginia Attn: Technical Library Div.
Commanding Officer Office of Naval Research Branch Office 495 Summer Street Boston 10, Mass.	Chief, Bur. of Naval Weapons Department of the Navy Washington 25, D.C. Attn: Codes R R-12 RR RRRE RU RUTO	Computation and Exterior Ballistics Lab. (Dr. Hershey) Commanding Officer NROTC and Naval Adm. Unit Massachusetts Inst. of Tech. Cambridge 39, Mass. Commanding Off. and Dir. Underwater Sound Laboratory Fort Trumbull New London, Conn. Attn: Technical Library
Commanding Officer Office of Naval Research Branch Office 230 No. Michigan Ave. Chicago 1, Illinois	Commanding Off. and Director David Taylor Model Basin Washington 7, D.C. Attn: Codes 142 500 513 521 526 550 563 589	Commanding Off. and Dir. U.S. Navy Mine Def. Lab. Panama City, Florida Superintendent U.S. Naval Postgrad. School Monterey, California Attn: Library
Commanding Officer Office of Naval Research Branch Office 207 West 24th Street New York 11, New York	Commander Naval Ordnance Laboratory Silver Spring, Maryland Attn: Dr. A. May Desk DA Desk HL Desk DR	Commanding Off. and Dir. Naval Electronic Lab. San Diego 52, California Attn: Code 4223 Commanding Off. and Dir. U.S. Naval Civil Eng. Lab. Port Hueneme, California
Commanding Officer Office of Naval Research Branch Office 1030 East Green Street Pasadena 1, Calif.	Commander Naval Ordnance Test Station China Lake, Calif. Attn: Mr. J.W. Hicks Codes 5014 4032 753	Mr. C. K. Chatten Code 949 New York Naval Shipyard Material Laboratory Brooklyn 1, New York Commander Norfolk Naval Shipyard Portsmouth, Virginia
Commanding Officer Office of Naval Research Branch Office Navy No. 100, Box 39 Fleet Post Office New York, New York (25)		
Commanding Officer Office of Naval Research Branch Office 1000 Geary Street San Francisco 9, California		
Director Naval Research Laboratory Washington 25, D. C. Attn: Codes 2000 2020 2027 (6)		
Chief, Bur. of Yards and Docks Department of the Navy Washington 25, D.C. Attn: Codes D-202 D-400 D-500	Superintendent U.S. Naval Academy Annapolis, Maryland Attn: Library	

Commander U.S. Naval Ord. Test Sta. Pasadena Annex 3202 E. Foothill Blvd. Pasadena, Calif. Attn: Mr. J. W. Hoyt Research Division P508 P804 P807 P80962 (Library Sec)	Commandant U.S. Coast Guard 1300 E Street, NW Washington, D. C. Beach Erosion Board U.S. Army Corps of Engrs. Washington 25, D. C. Commanding Officer U.S. Army Research Office (Box CM, Duke Station Durham, N. C.	Director Ames Research Laboratory National Aero and Space Adm. Moffett Field, Calif. Director Lewis Research Center National Aero and Space Adm. Cleveland, Ohio Director Engineering Science Division National Science Foundation Washington, D. C.
Commander New York Naval Shipyard Naval Base Brooklyn, New York	Commander Hdqs. U.S. Army Transportation Res. and Development Comm. Transportation Corps Fort Eustis, Virginia	Commander Air Force Cambridge Res. Ctr. 230 Albany Street Cambridge 39, Mass. Attn: Geophysical Res. Library
Commander Boston Naval Shipyard Boston 29, Mass.	Director U.S. Army Engineering Res. and Development Labs. Fort Belvoir, Virginia Attn: Tech. Documents Ctr.	Air Force Off. of Sci. Res. Mechanics Division Washington 25, D. C.
Commander Philadelphia Naval Shipyard Naval Base Philadelphia 12, Penn.	Office of Technical Services Dept. of Commerce Washington 25, D. C. Defense Documentation Ctr. Cameron Station Alexandria, Virginia (10)	National Research Council Montreal Road Ottawa 2, Canada Attn: Mr. E. S. Turner
Commander Portsmouth Naval Shipyard Portsmouth, N. H. Attn: Design Division	Maritime Administration 441 G Street, NW Washington 25, D. C. Attn: Coordinator of Research Div. of Ship Design	Engineering Societies Library 29 West 39th Street New York 18, N. Y. Society of Naval Architects and Marine Engineers 74 Trinity Place New York 6, N. Y.
Commanding Officer U.S. Naval Underwater Ord. Sta. Newport, Rhode Island Attn: Research Division	Fluid Mechanics Section National Bureau of Standards Washington 25, D. C. Attn: Dr. G. B. Schubauer	Webb Inst. of Naval Architecture Glen Cove, Long Island, N. Y. Attn: Prof. E. V. Lewis Technical Library
Commander Long Beach Naval Shipyard Long Beach 2, Calif.	U.S. Atomic Energy Commission Tech. Infor. Ser. Ext. P.O. Box 62 Oakridge, Tenn.	The Johns Hopkins University Baltimore 18, Maryland Attn: Prof. S. Corrsin Prof. F. H. Clauser Prof. O. M. Phillips
Commander Pearl Harbor Naval Shipyard Navy No. 128, Fleet Post Off. San Francisco, Calif.	Director of Research National Aero. and Space Adm. 1512 H Street, NW Washington 25, D. C.	Director Applied Physics Laboratory The Johns Hopkins Univ. 8621 Georgia Avenue Silver Spring, Maryland
Commander San Francisco Naval Shipyard San Francisco 24, Calif.	Director Langley Research Ctr. National Aero and Space Adm. Langley Field, Va.	New York State University Maritime College Engineering Department Fort Schuyler, New York Attn: Prof. J. J. Foody
Superintendent U.S. Merchant Marine Academy Kings Point Long Island, New York Attn: Dept. of Engineering		

California Inst. of Tech. Pasadena, California Attn: Hydrodynamics Lab. Prof. T. Y. Wu Prof. A. Ellis Prof. A. Acosta	Institute for Fluid Mechanics and Applied Mathematics University of Maryland College Park, Md. Attn: Prof. J.M. Burgers Cornell Aeronautical Lab. Buffalo 21, N. Y. Attn: Mr. W.F. Milliken, Jr.	Hamburgische Schiffbau- Versuchsanstalt Bramfelder Strasse 164 Hamburg 33, Germany Attn: Dr. O. Grim Max-Planck Institut fur Stromungsforschung Bottingerstrasse 6-8 Gottingen, Germany Attn: Dr. H. Reichardt, Dir.
University of California Berkeley 4, California Attn: Dept. of Eng Prof. H.A. Schade Prof. J. Johnson Prof. J.V. Wehausen Prof. E.V. Laitone Prof. P. Lieber Prof. M. Holt	Brown University Providence 12, R. I. Attn: Dr. R. E. Meyer Dr. W. H. Reid Stevens Inst. of Tech. Davidson Laboratory Hoboken, New Jersey Attn: Mr. D. Savitsky Mr. J.P. Breslin Dr. D.N. Hu Dr. S.J. Lukasik	Prof. Dr.-Ing. S. Schuster, Baudirector Versuchsanstalt fur Wasserbau und Schiffbau Berlin, Germany Netherlands Ship Model Basin Wageningen, The Netherlands Attn: Ir. R. Wereldsma Dr. J.B. van Manen
University of California Los Angeles, California Attn: Prof. R.W. Leonard Prof. A. Powell		
Director Scripps Inst. of Oceanography University of California La Jolla, California	Director Woods Hole Oceanographic Institute	Mitsubishi Shipbuilding and Engineering Company Nagasaki, Japan Attn: Dr. K. Taniguchi
Iowa Inst. of Hydraulic Res. State University of Iowa Iowa City, Iowa Attn: Prof. H. Rouse Prof. L. Landweber Prof. P.G. Hubbard	Woods Hole, Mass. Director Alden Hydraulic Lab. Worcester Poly. Inst. Worcester, Mass.	Mr. W. R. Wiberg, Chief Marine Performance Staff The Boeing Company Aero-Space Division P. O. Box 3707 Seattle 24, Washington
Harvard University Cambridge 38, Mass. Attn: Prof. G. Birkhoff Prof. S. Goldstein	Stanford University Stanford, California Attn: Dr. Byrne Perry Prof. P. Garabedian Mr. L.I. Schiff Dr. S. Kline Prof. E.Y. Hsu	Mr. William P. Carl Grumman Aircraft Corp. Bethpage, Long Island New York Mr. G. W. Paper ASW and Ocean Systems Dept. Lockheed Aircraft Corp. Burbank, California
University of Michigan Ann Arbor, Michigan Attn: Engineering Res. Inst.	Department of Theoretical and Applied Mechanics College of Engineering University of Illinois Urbana, Illinois Attn: Dr. J.M. Robertson	Dr. A. Ritter Therm Advanced Res. Div. Therm, Incorporated Ithaca, New York
Director St. Anthony Falls Hyd. Lab. University of Minnesota Minneapolis 14, Minn. Attn: Mr. J.N. Wetzel Prof. B. Silberman Prof. L.G. Straub	Department of Mathematics Rensselaer Poly. Inst. Troy, New York Attn: Prof. R.C. DiPrima Southwest Research Inst. 8500 Culebra Road San Antonio 6, Texas Attn: Dr. H.N. Abramson	Hydronautics, Incorporation Pindell School Road Howard County Laurel, Maryland Attn: Mr. P. Eisenberg, Pres. Mr. M.P. Tulin, V.P.
Massachusetts Inst. of Tech. Cambridge 39, Mass. Attn: Prof. P. Mandel Prof. M.A. Abkowitz	Dept. of Aero. Engineering University of Colorado Boulder, Colorado Attn: Prof. M.S. Uberoi	Dr. J. Kotik Technical Research Group, Inc. Route 110 Melville, New York AiResearch Mfg. Company 9851-9951 Sepulveda Blvd. Los Angeles 45, Calif. Attn: Blaine R. Parkin

Hydrodynamics Laboratory
Convair
San Diego 12, California
Attn: Mr. H. E. Brooke
Mr. R. H. Oversmith

Baker Manufacturing Company
Evansville, Wisconsin

Gibbs and Cox, Inc.
21 West Street
New York 16, New York

Electric Boat Division
General Dynamics Corp.
Groton, Connecticut
Attn: Mr. R. McCandliss

Armour Research Foundation
Illinois Inst. of Tech.
Chicago 16, Illinois
Attn: Library

Missile Development Div.
North Amer. Aviation, Inc.
Downey, California
Attn: Dr. E. R. Van Driest

National Physical Laboratory
Teddington, Middlesex, England
Attn: Head Aerodynamics Division
Mr. A. Silverleaf

Aerojet General Corporation
6352 Irwindale Ave.
Azusa, California
Attn: Mr. C. A. Gongwer

Astropower, Inc.
2121 Paularino Ave.
Newport Beach, Calif.
Attn: R. D. Bowerman

Transportation Tech. Res. Inst.
No. 1057-1-Chome
Mejiro-machi, Toshima-ku
Tokyo-to, Japan

Oceanics, Incorporated
Plainview, Long Island, N. Y.
Attn: Dr. Paul Kaplan

Prof. C. S. Yih
University of Michigan
Ann Arbor, Michigan

Grumman Aircraft Corp.
Bethpage, Long Island, N. Y.
Attn: Eng. Lib, Plant 5
Mr. Leo Geyer

Professor J. William Holl
Garfield Thomas Water Tunnel
Ordnance Research Laboratory
The Pennsylvania State Univ.
P. O. Box 30
State College, Penn.

Professor George Carrier
Harvard University
Cambridge 38, Mass.

Dr. E. R. G. Eckert
Mechanical Eng. Dept.
University of Minnesota
Minneapolis, Minn. 55455

UNCLASSIFIED

Security Classification

DOCUMENT CONTROL DATA - R&D

(Security classification of title, body of abstract and indexing annotation must be entered when the overall report is classified)

1. ORIGINATING ACTIVITY (Corporate author) California Institute of Technology Division of Engineering and Applied Science		2a. REPORT SECURITY CLASSIFICATION UNCLASSIFIED	
		2b. GROUP Not Applicable	
3. REPORT TITLE STABILITY OF A CONDUCTING FLUID FLOWING DOWN AN INCLINED PLANE IN A MAGNETIC FIELD			
4. DESCRIPTIVE NOTES (Type of report and inclusive dates) Technical Report			
5. AUTHOR(S) (Last name, first name, initial) Hsieh, Din-Yu			
6. REPORT DATE December, 1964		7a. TOTAL NO. OF PAGES 19	7b. NO. OF REFS 9
8a. CONTRACT OR GRANT NO. NONR 220(28)		9a. ORIGINATOR'S REPORT NUMBER(S) Report No. 85-29	
b. PROJECT NO. NR-062-059		9b. OTHER REPORT NO(S) (Any other numbers that may be assigned this report) - - -	
c.			
d.			
10. AVAILABILITY/LIMITATION NOTICES Qualified requesters may obtain copies of this report from DDC.			
11. SUPPLEMENTARY NOTES		12. SPONSORING MILITARY ACTIVITY Office of Naval Research	
13. ABSTRACT A stability analysis is made for the laminar flow of a layer of a viscous and electrically conducting fluid down an inclined plane in a transverse magnetic field. It is found that the effect of the magnetic field, revealed through the Hartmann number, is to stabilize the flow. A simpler and physically clearer approximate treatment of the same problem based on the principle of local balance is also given. The results agree quite satisfactorily with the exact analysis.			

14.

KEY WORDS

Hydromagnetic stability
 Fluid Physics
 Laminar flow of a viscous fluid

LINK A

LINK B

LINK C

ROLE

WT

ROLE

WT

ROLE

WT

INSTRUCTIONS

1. **ORIGINATING ACTIVITY:** Enter the name and address of the contractor, subcontractor, grantee, Department of Defense activity or other organization (*corporate author*) issuing the report.

2a. **REPORT SECURITY CLASSIFICATION:** Enter the overall security classification of the report. Indicate whether "Restricted Data" is included. Marking is to be in accordance with appropriate security regulations.

2b. **GROUP:** Automatic downgrading is specified in DoD Directive 5200.10 and Armed Forces Industrial Manual. Enter the group number. Also, when applicable, show that optional markings have been used for Group 3 and Group 4 as authorized.

3. **REPORT TITLE:** Enter the complete report title in all capital letters. Titles in all cases should be unclassified. If a meaningful title cannot be selected without classification, show title classification in all capitals in parenthesis immediately following the title.

4. **DESCRIPTIVE NOTES:** If appropriate, enter the type of report, e.g., interim, progress, summary, annual, or final. Give the inclusive dates when a specific reporting period is covered.

5. **AUTHOR(S):** Enter the name(s) of author(s) as shown on or in the report. Enter last name, first name, middle initial. If military, show rank and branch of service. The name of the principal author is an absolute minimum requirement.

6. **REPORT DATE:** Enter the date of the report as day, month, year; or month, year. If more than one date appears on the report, use date of publication.

7a. **TOTAL NUMBER OF PAGES:** The total page count should follow normal pagination procedures, i.e., enter the number of pages containing information.

7b. **NUMBER OF REFERENCES:** Enter the total number of references cited in the report.

8a. **CONTRACT OR GRANT NUMBER:** If appropriate, enter the applicable number of the contract or grant under which the report was written.

8b, 8c, & 8d. **PROJECT NUMBER:** Enter the appropriate military department identification, such as project number, subproject number, system numbers, task number, etc.

9a. **ORIGINATOR'S REPORT NUMBER(S):** Enter the official report number by which the document will be identified and controlled by the originating activity. This number must be unique to this report.

9b. **OTHER REPORT NUMBER(S):** If the report has been assigned any other report numbers (*either by the originator or by the sponsor*), also enter this number(s).

10. **AVAILABILITY/LIMITATION NOTICES:** Enter any limitations on further dissemination of the report, other than those

imposed by security classification, using standard statements such as:

- (1) "Qualified requesters may obtain copies of this report from DDC."
- (2) "Foreign announcement and dissemination of this report by DDC is not authorized."
- (3) "U. S. Government agencies may obtain copies of this report directly from DDC. Other qualified DDC users shall request through _____."
- (4) "U. S. military agencies may obtain copies of this report directly from DDC. Other qualified users shall request through _____."
- (5) "All distribution of this report is controlled. Qualified DDC users shall request through _____."

If the report has been furnished to the Office of Technical Services, Department of Commerce, for sale to the public, indicate this fact and enter the price, if known.

11. **SUPPLEMENTARY NOTES:** Use for additional explanatory notes.

12. **SPONSORING MILITARY ACTIVITY:** Enter the name of the departmental project office or laboratory sponsoring (*paying for*) the research and development. Include address.

13. **ABSTRACT:** Enter an abstract giving a brief and factual summary of the document indicative of the report, even though it may also appear elsewhere in the body of the technical report. If additional space is required, a continuation sheet shall be attached.

It is highly desirable that the abstract of classified reports be unclassified. Each paragraph of the abstract shall end with an indication of the military security classification of the information in the paragraph, represented as (TS), (S), (C), or (U).

There is no limitation on the length of the abstract. However, the suggested length is from 150 to 225 words.

14. **KEY WORDS:** Key words are technically meaningful terms or short phrases that characterize a report and may be used as index entries for cataloging the report. Key words must be selected so that no security classification is required. Identifiers, such as equipment model designation, trade name, military project code name, geographic location, may be used as key words but will be followed by an indication of technical context. The assignment of links, roles, and weights is optional.